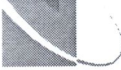


1. Approving Civil Aviation Authority Country: 2.		3. Form Tracking Number: 518039-236254-614	
FAA/United States 		AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG	
4. Organization Name and Address: CRANE AEROSPACE & ELECTRONICS ELDEC CORPORATION 16700 13th Avenue West Lynnwood, WA 98037-8597 (AG4R049M)		5. Work Order/Contract/Invoice Number: WR2171538	
6. Item:	7. Description:	8. Part Number:	9. Quantity:
1	PCM, FCE	4-956-02 (4091679-901)	1
12. Remarks		10. Serial Number:	11. Status/Work:
TSN: N/A CSN: N/A TSO: N/A CSO: N/A Unit passed isolation test and functional incoming test. Unit passed visual inspection and final test; processed in accordance with CMM 27-02-02, rev 6; dated: Oct/28/16. Honeywell PO#: 4207715103. Honeywell P/N: 4091679-901. Honeywell S/N: 236254-614. Service Order#: 005011806067. Certifies that the work specified in Block 11/12 was carried out in accordance with EASA Part 145 and in respect to that work the article is considered ready for release to service under EASA Part 145 approval no. EASA 145-5067.		236254-614	Inspected
13a. Certifies that items identified above were manufactured in conformity to ☐ Approved design data and are in a condition for safe operation ☐ Non-approved design data specified in Block 12		14a. ☒ 14 CFR 43.9 Return to Service ☐ Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43, and in respect to that work, the items are approved for return to service.	
13b. Authorized Signature:	13c. Approval/Authorization No:	14c. Approval/Certificate No: AG4R049M	
13d. Name (Typed or Printed)	13e. Date (dd/mm/yy)	14d. Date (dd/mm/yy): 20/DEC/2017	
User/Installer Responsibilities			
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1 it is essential that the user/installer ensure that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.			
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.			

FAA Form 8130-3 (02-14)

Printed on 20-DEC-2017 at 08:43:26

NSN-0052-00-012-9005

**ELDEC CORPORATION**16700 13th Avenue West
Lynnwood, WA 98037-8597

Run Date: Dec 20, 2017 4:43 PM

RMA #

518039-236254-614**WARRANTY & REPAIR STRIP REPORT**Phone# (425)743-8255 (Product Support Admin)
24 AOG # (425) 743-8100**2017/12/20 This document contains Data controlled by the EAR. Diversion or use contrary to U.S. law is prohibited. ECCN: 9E991**

RMA NUMBER	PART NUMBER	AS RCVD	SERIAL NUMBER	AS RCVD	RECEIVED DATE	RECEIVED BY	QTY
518039-236254-614	4-956-02		236254-614		Dec 14, 2017	269300	1
CUSTOMER	CUSTOMER PN		CUSTOMER S/N	HOURS		AIRCRAFT ID	
HONEYWELL INTERNATIONAL INC	4091679-901				TSN		
SERVICE RQST#				CYCLES			
858529					CSN		
DATE OF MFG	CUSTOMER PO	REJECTION REF NO.	FAA 8130 STATUS	FAILURE CLASS	REJ VER?		
May 1, 2012	4207715103		Inspected	NA	N		
MOD LEVEL RCVD	MOD LEVEL SHIPPED	SHIPPED PART#	SHIPPED SERIAL #				
A	A	4-956-02	236254-614				

HISTORY OF UNIT

RMA NUMBER	RECEIVE DATE	CUSTOMER	REJ VER?	PROB TYPE	CUSTOMER PO NUMBER
518039-236254-614	Dec 14, 2017	HONEYWELL INTERNATIONAL INC	N	NFF	4207715103

CUSTOMER REJECTION DESCRIPTION**A23 - ADV MSG SPOILER APPEARS DURING CREW CHK.
QUOTE UNIT FOR REPAIR DIRECTIONS. INCORPORATE ALL APPLICABLE MODs.****TROUBLESHOOTING AND WORK PERFORMED****Customer rejection: not verified. Unit received with seal intact. Minor scratches on chassis; no obvious damage. Inspected connectors/pins; normal. Unit passed isolation test and functional test. Inspected CCAs; no obvious damage. Unit passed visual inspection and final test; processed in accordance with CMM 27-02-02, rev 6; dated: Oct/28/16.****PARTS USED**

Part Number	Description	Quantity Issued
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