| TUNESS APPROVAL TAG  | 1 CP-2B,AIRBUS DUAL- 1146000-102 1 SYSTEM  | EASA Part 145 Approval Number EASA.145.4618.   | in conformity to:    14 CFR 43.9 Return  | esign data specified in Block 12:  13c. Approval/Authorization No.:  14b. Authorized Signature:  | Organization Name and Add Organization Name and Add  T. Description CP-2E  T. Remarks: REPAIRED AND TESTED PER HONEYWELL P/N 930-6201- CUSTOMER PO# T42330 TSN / CSN : 2513 : 37 / 399   | FAA Form 8130-3, AIRWORTHINESS APPROVAL  | 146AA0568   | Work Order/Contract/Invoice Number: 3505643265  1. Status/Work: REPAIRED  |
|--|--|--|--|--|--|--|---|---|
|  | 11910 BEVERLY PARK ROAD  11910 BEVERLY PARK ROAD  EVERETT, WA 98204  9. Quantity:  10. Serial Number:  11. Status/Work   | 1   10   10   10   10   10   10   10   | To Description:   To Descrip   | 113 0 BEFERT WA 9204   10. Serial Number   11. Status/Work   REPAIRED  | FAA/UNITED STATES  | dress:   | 5   | 3505643265  |
| FAA Form 8130-3, AIRWOK I TIINCO   |  | 1 CP-2B,AIRBUS DUAL- SYSTEM  146000-102  1 CP-2B,AIRBUS DUAL- SYSTEM  1 CP | Item: 1 CP-2B,AIRBUS DUAL-SYSTEM  1 CP-2B,AIRBUS DUAL-SYST | CP-2B,AIRBUS DUAL- SYSTEM  1146000-102  CP-2B,AIRBUS DUAL- SYSTEM  1146000-102  11460000-102  11460000-102  11460000000000000000000000000000000000   |  | EVERETT, WA 98204  [8. Part Number:  | Serial Number: 11.  | REPAIRED  |
| A/UNITED STATES  FAA Form 8130-3, AIRWOK ITIMED  KORRY ELECTRONICS COMPANY 11910 BEVERLY PARK ROAD Organization Name and Address: 11910 BEVERLY PARK ROAD EVERETT, WA 98204  7. Description:   |  | Remarks:  (EPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  (ONEYWELL P/N 930-6201-001 S/N AA0568.  (CUSTOMER PO# T42330  (SON : 2513 : 37 / 399)  (CSN : 2513 : 37 / 399)  | 2. Remarks:  REPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  REPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  REPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  REPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  REPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  REPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  REPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  REPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  REPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  REPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  REPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  REPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  REPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  REPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  REPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  REPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  REPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  REPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  REPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  REPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  REPAIRED AND TESTED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  REPAIRED AND TESTED AND TESTED AND TESTED AND TESTED AND TESTED AND T | D AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63  FELL P/N 930-6201-001 S/N AA0568.  ER PO# T42330  N: 2513: 37 / 399  N: 2513: 37 / 399  Liter work specified in Blocks 11/12 was carried out in accordance with EASA Part 145 and in respect to that work 145 Approval Number EASA.145.4618.  Liter work specified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work specified in accordance with EASA Part 145 and in respect to that work 145.45 Approval Number EASA.145.4618.  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufactured in conformity to:  Liter work identified above were manufacture |  |  |   |   |
| A/UNITED STATES  FAA Form 8130-3, AIRWOK ITIMES  KORRY ELECTRONICS COMPANY 11910 BEVERLY PARK ROAD  EVERETT, WA 98204  7. Description:   |  | TSN / CSN : 2513 : 37 / 399  Constitute that the work specified in Blocks 11/12 was carried out in accordance with EASA Part 145 and in respect to that work the product/article is considered ready for release to service under the work specified in Blocks 11/12 was carried out in accordance with EASA Part 145 and in respect to that work the product/article is considered ready for release to service under the work specified in Blocks 11/12 was carried out in accordance with EASA Part 145 and in respect to that work the product/article is considered ready for release to service under the work specified in Blocks 11/12 was carried out in accordance with EASA Part 145 and in respect to that work the product/article is considered ready for release to service under the work specified in Blocks 11/12 was carried out in accordance with EASA Part 145 and in respect to that work the product/article is considered ready for release to service under the work specified in Blocks 11/12 was carried out in accordance with EASA Part 145 and in respect to that work the product/article is considered ready for release to service under the work specified in Blocks 11/12 was carried out in accordance with EASA Part 145 and in respect to that work the product/article is considered ready for release to service under the production of the product | 3d in Blocks 11/12 was carried out in accordance with EASA Part 145 and in respect to that work the product/articl   | nd in Blocks 11/12 was carried out in accordance with EASA Part 145 and in respect to that work the reasonable to the secondance with EASA Part 145 and in respect to that work the reasonable to the respect to that work the secondary to the respect to that work the secondary that we were manufactured in conformity to:    14a.     14b.     14c.     14c.  | 2. Remarks:<br>REPAIRED AND TESTED PI<br>HONEYWELL P/N 930-6201-   | PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA003572<br>01-001 S/N AA0568.  | -1. TSO-C63c.   |   |
| NITED STATES   | D AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 EELL P/N 930-6201-001 S/N AA0568. ER PO# T42330 N : 2513 : 37 / 399   | Continue that the work specified in Blocks 11/12 was continued.  | Certifies that the work specified in Blocks 1112 was believed.  EASA Part 145 Approval Number EASA.145.4618.   | Certifies that the work specified in Blocks 1712 was secured in Conformity to:  Certifies the tire tiems identified above were manufactured in conformity to:  Certifies the tire tiems identified above were manufactured in conformity to:  Certifies the tire tiems are approved design data and are in a condition for safe operation.  Approved design data and are in a condition for safe operation.  The tiems are approved for return to service.  Approved design data and are in a condition for safe operation.  The tiems are approved for return to service.  Work, the items are approved for return to service.  | ON CON   |  |   |   |
| NITED STATES  KORRY ELECTRONICS COMPANY (KE7R393J)  FAA Form 8130-3, AIRWOK ITIMALOUS  KORRY ELECTRONICS COMPANY (KE7R393J)  FOR THE PARK ROAD  EVERETT. WA 98204  9. Quantity: 10. Serial  1. Part Number 1146000-102  SYSTEM  DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63  DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63  ELL P/N 930-6201-001 S/N AA0568.  ER PO# 142330  N: 2513: 37 / 399  N: 2513: 37 | D AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 DAND TESTED PER CMM 34-41-83 REV 5 DATED 14 D | Certifies that unless otherwise was accomplished in accordan work, the items are approved for the Authorized Signature:  14b. Authorized Signature:  | esign data specified in Black 12:  13c. Approval/Authorization No.:  14b. Authorized Signature:  15c. Approval/Authorization No.:  |  | Certifies that the work specified EASA Part 145 Approval Numbe  13a.  Certifies the items identified above design data and approved design data approved design  | in Blocks 11/12 was carried out in accordance with EASA Part 145 er EASA.145.4618.  were manufactured in conformity to:  at a specified in Block 12.  13c. Approval/Authorization No.:   | t to that work the product/article is consider  CFR 43.9 Return to Service  CFR 43.9 Return to Service of Pedera  shed in accordance with Tritle 14, Code of Federa  s are approved for return to service.  ed Signature:   | gred ready for release to service unde gulation specified in Block 12 identified in Block 11 and described in Block 12 al Regulations, part 43 and in respect to the late of the second |
| NITED STATES  KORRY ELECTRONICS COMPANY 11910 BETRONICS COMPANY 11910 BETRONIC | D AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63 ELL P/N 930-6201-001 S/N AA0568. ER PO# T42330 N : 2513 : 37 / 399 N : 2513 : 37 /  | Certifies that unless otherwise was accomplished in accordan work, the items are approved for 13e. Date (dd/mmm) (14d. Name (Typed or Printed 14d. Name (Typed or Printed  | esign data specified in Plack 12.  13c. Approval/Authorization No.:  14b. Authorized Signature:  14c. Approval/Cer  14c. Approval/Cer  14d. Name (Typed or Printed):  14e. Date (dd/mrn  | nted)  13e. Date (dd/mmm/yyyy):  14d. Name (Typed or Printed):  14e. Date (dd/mr 1   | Certifies that the work specified in EASA Part 145 Approval Number Certifies the items identified above Certifies above Certifies the items identified above Certifies above Certifies the items identified above Certifies above Certifies the items identified above Certifies above Cert | in Blocks 11/12 was carried out in accordance with EASA Part 145 er EASA.145.4618.  were manufactured in conformity to:  ata specified in Block 12-  13c. Approval/Authorization No.:  | creating to that work the product/article is considered to that work the product/article is considered.  Creating to Service of Code of Federa thed in accordance with Title 14, Code of Federa as are approved for return to service.  Typed or Printed):  Vivian Tu | gred ready for release to service under gred ready for release to service under the service under the service in Block 12 gulation specified in Block 11 and described in Block 12 and in respect to the service with the service under |
| Aduntified States    FAA Form 8130-3, AIRWOR   Interview   State   Sta | Examples and the work specified in Blocks 11/12 was carried out in accordance with EASA Part 145 and in respect to that work the productianticle is considered ready for release to service und Continuous Poperation Style (1908). Part 145 Approved design data specified in Block 12 (30. Approval Authorization No. 145. Approval data specified in Block 12 (30. Approval Authorization No. 145. Approval data specified in Block 12 (30. Approval Authorization No. 145. Approval data specified in Block 12 (30. Approval Authorization No. 145. Approval data specified in Block 12 (30. Approval Authorization No. 145. Approval data specified in Block 12 (30. Approval Authorization No. 145. Approval (Authorization No. 145. Approval (Certificate No. 146. Approval (Certificate No. 146. Approval (Certificate No. 146. Approval (Certificate No. 146. Base (dd/mmm/ywy)) (146. Base (dd/mm/mm/ywy)) (146. Base (dd/mm/mm/ywy)) (146. Base (dd/mm/mm/ywy)) (146. B | Certifies the items identified above were manufactured in conformity to:  Certifies the titems identified above were manufactured in conformity to:  Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12.  Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12.  Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12.  Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12.  Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12.  Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12.  Certifies that unless otherwise specified in Block 12, the work identified in Block 12 and in respect to was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to work, the items are approved for return to service.  ME7/R393J  13b. Authorized Signature:  14c. Approval/Certificate No.:  KE7R393J  13d. Name (Typed or Printed):  14d. Name (Typed or Printed):  | 13b. Authorized Signature:  13c. Approval/Authorization No.:  13b. Authorized Signature:  13c. Approval/Authorization No.:  14b. Authorized Signature:  14c. Approval/Certificate No.:  KE7R393J  13d. Name (Typed or Printed):  14d. Name (Typed or Printed):  14d. Name (Typed or Printed):  11de. Date (dd/mmm/yyyy)  11de. Date (dd/mmm/yyyy)  11de. Date (dd/mmm/yyyy)  11de. Date (dd/mmm/yyyy)  | 13e. Date (dd/mmm/yyyy): 13d. Name (Typed or Printed): 13e. Date (dd/mmm/yyyy): 11d. Name (Typed or Printed): 11d. Name (Typed | Certifies that the work specified in EASA Part 145 Approval Number Certifies the terms identified above Certifies the terms identified above Dentifies the terms identified above design data and the terms identified above data and the terms identified  | ed in Blocks 11/12 was carried out in accordance with EASA Part 145 and in responder EASA.145.4618.  The manufactured in conformity to:  The properties of the interval of the | t to that work the product/article is consider CFR 43.9 Return to Service CFR 43.9 Return to Service nless otherwise specified in Block 12, the work in accordance with Title 14, Code of Federas are approved for return to service.  Typed or Printed):  Vivian Tu  | gred ready for release to service und gulation specified in Block 12 identified in Block 11 and described in Bal Regulations, part 43 and in respect to 4c. Approval/Certificate No.:  KE7R393J  14e. Date (dd/mmm/yyyy)  11 Dec 2023   |
| AUUNITED STATES  RAA Form 8130-3, AIRWURITED STATES  Copporation Name and Address:  Instit BECRET PARK ROAD  SOB643265  REPAIRED  11910 BECRET PARK ROAD  11910 BECRET PARK ROAD  11910 BECRET PARK ROAD  12 Description:  1 CP-2B AIRBUS DUAL-  SYSTEM  1 CP-2B AIRBUS DUAL-  1 CP-2B AIRBUS DUAL-  SYSTEM  1 AS and in respect to final work the product/article is considered ready for release to service under the system of  | D AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63  DELL P/N 930-6201-001 S/N AA0568.  ELL P/N 930-6201-001 S/N AA0568.  ELL P/N 930-6201-001 S/N AA0568.  ER PO# T42330  N: 2513: 37 / 399  Non-approved design data specified in example out in accordance with EASA Part 145 and in respect to that work in accordance with the national regulations of an amounthiness authority of the country specified for the aircraft manthemarker authority accepts aircraft engine (s) propeller(s) propeller(s) from the aircraft manthemarker accords must contain an instal hat his/her aircraft manthemarker accords must contain an instal hat his/her aircraft are giness aircraft engine (s) propeller(s) from the aircraft engine that his/her aircraft engine succepts aircraft engine (s) propeller(s) from the aircraft engine to contain an instal hat his/her aircraft engine succepts aircraft engine (s) propeller(s) aircraft engin | Certifies Inestigms identified above were manufactured in conformity to:  Certifies Inestigms identified above were manufactured in conformity to:  Certifies Inestigms identified above were manufactured in conformity to:  Approved design data and are in a condition for safe operation.  Approved design data specified in Block 12.  Non-approved design data specified in Block 12.  Approval/Certificate No.:  (Certifies that unless otherwise specified in Block 12, the work identified in Block 12 the work in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to was accomplished in Block 12.  At CFR 43.9  | 13c. Approval/Authorization No.: 13b. Authorized Signature: 13c. Approval/Authorization No.: 13d. Name (Typed or Printed): 14d. Name (Typed or Printed): 14d | 13e. Date (dd/mmm/yyx):  13e. Date (dd/mmm/yyx):  13e. Date (dd/mmm/yxx):  14e. Date (dd/mmm/yxx):  17e. Date (dd/mmm/yxx | Certifies that the work specified in EASA Part 145 Approval Number Certifies the items identified above design data and items identified above d | ed in Blocks 11/12 was carried out in accordance with EASA Part 145 and in responder EASA.145.4618.  14a.  2 1  14a.  14b. Authorization No.  14b. Authorization No.  14c. Approval/Authorization No.  14d. Name  User/Installer Responsibilities  User/Installer Responsibilities  14d. Name  15d. Approvalidations of an airworthiness authority different the existence of this document alone does not automatically constitute authority different the same aircraft maintenance records muless authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority muless authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority muless authority accepts aircraft engine(s)/article(s) from the airworthiness authority muless authority accepts aircraft engine(s)/article(s) from the airworthiness authority muless authority accepts aircraft engine(s)/article(s) from the airworthiness authority muless authority accepts aircraft engine(s)/article(s) from the airworthiness authority muless authority accepts aircraft engine(s)/article(s)/article(s) from the airworthiness authority muless authority accepts aircraft engine(s)/article(s)/article(s) from the airworthiness authority muless authority accepts aircraft engine(s)/article(s)/ar | CFR 43.9 Return to Service CFR 43.9 Return to Service Other region accordance with Title 14, Code of Federa is are approved for return to service.  Typed or Printed):  Vivian Tu  12  14  15  16  17  17  17  17  17  17  18  19  19  19  19  10  10  10  10  10  10 | gred ready for release to service unc<br>gulation specified in Block 12<br>identified in Block 11 and described in It<br>identified in Block 11 and described in It<br>al Regulations, part 43 and in respect to<br>al Regulations, part 43 and in respect to<br>al RETR393J<br>(KETR393J<br>14e. Date (dd/mmm/lyyyy)<br>114e. Date (dd/mmm/lyyyy)<br>114e. Date (dd/mmm/lyyyy)<br>114e. Date (dd/mmm/lyyyy)<br>114e. Date (dd/mmm/lyyyy)<br>114e. Date (dd/mmm/lyyyy)<br>114e. Date (dd/mmm/lyyyy)   |

| <b>苏</b> 活田体   | <b>种田头</b> 将型形积它日子运  |   |                   |  |   | Name (FIIIIceu)  |
|--|--|---|-------------------|--|---|--|
|  |  |   |                   | Vivian Tu  | (打印的)   | 1  |
|  | 0100872  | MOC No. F00100872   |                   |  |   | Signature  |
|  | 19 中国民航总局授权<br>Issued by or on behalf of the CAAC  | 19 中国民航总局授权<br>Issued by or on beh  | 11-Dec-2023       | 18 批准日期<br>Date 11   |   | 16 批准人签名   |
| 兹声明上述产品除第13项的其它规定以外,已接照上述国家追肌杂例和应口自应不兹声明上述产品除第13项的其它规定以外,已接照上述国家追肌杂例和应口自应不在数据,这个品处于安全可用状态可以批准放行使用。<br>Certifies that the work specified above except as specified in block 13<br>Certifies that the work specified above except as specified in block 13<br>was carried out in accordance with the airworthiness regulations of the<br>was carried out in accordance with the airworthiness regulations of the<br>stated country and the notified special requirements of the importing<br>stated country and in respect to that work, the part(s) is (are) in condition<br>country and in respect to that work, the part(s) is (are) in condition<br>for safe operation and considered ready for release to service. (over) | 除第13项的其它规定以外,<br>了工作,该产品处于安全<br>计he work specified ab<br>t in accordance with t<br>and the notified spec<br>and the to that work,<br>respect to that work,<br>tion and considered re | 兹声明上述产品版<br>兹声明上述产品版<br>的特殊要求进行。<br>Certifies that<br>was carried out<br>stated country<br>country and in<br>for safe opera |                   | 照上述国家适航条例进行制造号设计资料和进口国提出的专会 except as otherwise spece except as otherwise the airworthing the case of parts to be fied special requirements  | 14 新产品 New Parts 兹声明上述产品烧箕13项的其它规定以外,已按照上述国家适航条例进行制造/兹声明上述产品烧箕13项的其它规定以外,已按照上述国家适航条例进行制造/检查,并且该产品(出口产品)符合经批准的型号设计资料和进口国提出的专用要求。 Certifies that the Part(s) identified above except as otherwise specified in block case with the airworthiness 13 was(were) manufactured/inspected in accordance with the airworthiness regulations of the stated country and/or in the case of parts to be exported with the approved design data and with the notified special requirements of the importing country. | 14 新文品 New Parts<br>兹声明上述产品除籍1<br>检查,并且该产品(比<br>Certifies that the<br>13 was(were) manufa<br>regulations of the<br>the approved design<br>importing country. |
| 计同个法数分向打进门围锚钩  | Used Parts   | 15、体用社的产品   |                   |  |   |  |
|  |  |   |                   |  | : 37 / 399  | TSN / CSN : 2513 : 37 / 399  |
|  |  |   |                   |  | T42330  | CUSTOMER PO# T42330  |
|  |  |   |                   |  | HONEYWELL P/N 930-6201-001 S/N AA0568.  | HONEYWELL P/N  |
|  | SC.  | 723-1. TSO-C63  | REF JO RMA0035    | REV 5 DATED 14 DEC 2021  | 13 备注 Remarks<br>REPAIRED AND TESTED PER CMM 34-41-83 REV 5 DATED 14 DEC 2021 REF JO RMA0035723-1. TSO-C63c.  | 13 备注 Remarks<br>REPAIRED AND  |
|  |  |   |                   |  |   |  |
|  |  |   | TYPES             |  |   |  |
|  |  | ,   | TO MULTIPLE       | 1146000-102  | CP-2B, AIRBUS DUAL-SYSTEM   | 1  |
| REPAIRED   | Serial/Batch No.   | Qty   | Eligibility       | Part No.   | 7 內容<br>Description   | 6 序号<br>Item   |
| 12 产品次念<br>Status/Work   | 11 系列号/批号  | 10 数量   | 9 适用性             | 8 年号   | EVERETT, WA 98204 USA   |  |
|  |  |   |                   |  | 11910 BEVERLY PARK ROAD   | 1  |
| 1V01Ce   | Work Order/Contract/Invoice  |   |                   |  | ion<br>KORRY ELECTRONICS COMPANY  | 4 单位 Organization<br>KOR   |
|  | 5 工作单/合同单/货单   |   | TIF ICAID/ AIM ON | AUTHORIZED RELEASE CERTIFICATE/ MINISTRUMENTAL STATEMENT |   | CIIII  |
| RMA0035723-1   | L TAG  | 淮标签<br>HINESS APPROVA   | 批准放行证书/适航批准标签     | 中国民用航空总局 CAAC LI 符合性<br><b>批准放行</b>  | 2.  | 1 国家 Country   |
| 3 证书编号 Certificate Ref. No.  |  | X 适航性 Airworthiness   |                   | 7  |   |  |
|  |  |   |                   |  |   |  |

## 批准放行证书/适航批准标签

AUTHORIZED RELEASE CERTIFICATE/AIRWORTHINESS APPROVAL TAG使用者/安装者职责

# USER/INSTALLER RESPONSIBILITIES

- (1) 必须明确: 本文件并不批准零件/组件/部件可以装到有关产品上。(5) 当庙田老/安班老庙田孫昌宁七四天公司
- 在国的适航当局能接受所指国家适航当局批准出口的零件/组件/部件。 (2)当使用者/安装者使用的是所在国适航当局的条例,而不是本表第 1 项中所指国家适航当局的条例时,使用者/安装者必须保证所
- (3)表中第 14 项、第 15 项的陈述,并不说明本表是安装批准。在所有情况下,航空器使用前,航空器使用者/安装者应把按本国适 航条例颁发的安装批准放入维修记录中。
- (1) It is important to understand that the existence of this document alone does not automatically constitute authority to install the
- from the Airworthiness Authority of the country specified in block 1. the country specified in block 1 it is essential that the user/installer ensure that his/her Airworthiness Authority accepts parts/components/assemblies (2) Where the user/installer works in accordance with the national regulations of an Airworthiness Authority different than the Airworthiness Authority of
- certification issued in accordance with the national regulation by the user/installer before the aircraft may be flown. (3) Statements 14 and 15 do not constitute installation certification. In all cases the aircraft maintenance record must contain an installation

AAC-038(12/94)

## **Korry** Repair Action Report

Customer Name: HONEYWELL INTERNATIONAL INC.

RMA RMA0035723-1 Number:

**Product Code** 

Quantity

**Korry Part** 

**Korry Part Description** 

Customer PO

**Customer Part** 

**Customer Reject** 

1210F

1146000-102

CP-2B, AIRBUS DUAL-SYSTEM 3505643265

930-6201-001

5017988224

Korry Serial No: 1146AA0568

Customer Serial No: AA0568

Date Code: 042014R

As Received Condition Type

Used

### **Shop Findings**

### **Customer Rejection:**

PLEASE QUOTE TO TEST AND RECERTIFY. PLEASE QUOTE TO REPAIR IF IT FAILS TEST.

REMOVED IN SERVICEABLE CONDITION FROM A/C REGN: A6-EYL

REMOVAL DATE 05 JUNE 2023

PLEASE RECORD THE COMPONENT TSN 2513:37 CSN 399 ON YOUR RELEASE CERTIFICATION

\*

QUOTE REQUIRED BEFORE REPAIR IF ANY CUSTOMER-INDUCED DAMAGE (CID) ACCEPTANCE TEST REPORT/REPAIR ACTION REPORT REQUIRED 8130-3 (EASA) REQUIRED CAAC REQUIRED

### As Received Condition:

-Article received at rev E. MOD 1 marked on Honeywell UID label. #24208.

-Light plate and all knobs have dirt. Need cleaning. #24208.

- -DZUS fasteners have dirt and paint chip. Need cleaning and touch up paint. #24208.
- -Mounting plate has paint chip around the edges. Needs touch up paint. #24208.
- -Housing has dirt and scratches. Needs cleaning and touch up paint. #24208.
- -Connectors have ESD caps. #24208.

### **Evaluation Findings:**

-Article was tested for function per CMM 34-41-83, rev 5; it passed. #24208.

-All rotary switches & encoders had good detents and functioned properly in systems 1 & 2. #24208.

-"GAIN" potentiometer functioned properly in systems 1 & 2. NOTE: Visual inspection found gasket on potentiometer was cracked. Potentiometer replacement required. #24208.

-Light plate illuminated evenly. #24208.

- -Article passed TVS Protection Device & Ground Bonding Resistance Tests. #24208.
- -Rework and test per CMM 34-41-83, rev 5, dated 14 Dec. 2021. #24208.
- -Remove and replace "GAIN" potentiometer and extracted screws. #24208.
- -Re-assemble article. #24208.
- -Clean and touch up paint as necessary. #24208.

| Name        | Completion Date | Cycles Since Installation (CSI) | Time Since Installation (TSI) |
|-------------|-----------------|---------------------------------|-------------------------------|
| JDELGADILLO | 12:00:00 AM     | N/A                             | N/A                           |

### **Quality Section**

### Discrepancy:

-Cracked gasket on "GAIN" potentiometer. #24208.

### Cause

-Wear and tear. #24208.

Printed: 12/11/2023 9:00:40 AM

### **Corrective Action:**

-N/A. #24208.

Order Type REW Name LBOURTAL

**Shop Action** 

Page 1 of 2

### **Action Description:**

- -Rework and test per CMM 34-41-83, rev 5, dated 14 Dec. 2021. #47645.
- -Removed and replaced "GAIN" potentiometer and extracted screws. #47645. -Re-assemble article. #47645.
- -Cleaned and touch up paint as necessary. #47645.

### Testing per CMM required

Note: Creation or submittal of ATP test data is according to the ATP or applicable inspection alerts or PO requirements.

### **Special Requirements:**

-Article to be shipped at Rev E. #24208.

Name LBOURTAL

The information contained herein is non-technical data. Customer Returns / Repair Action Report