

1.	DIRECTION GENERALE DE L'AVIATION CIVILE FRANCE		2.		AUTHORISED RELEASE CERTIFICATE Certificat Libérateur Autorisé EASA FORM 1 Formulaire 1 de L'EASA		3. Form Tracking Number N° de repère du Formulaire 87563 SO: 5464810 / 1.000		
4. Organisation Name and Address: Nom et Adresse de l'Organisme :		REVIMA 1, AVENUE DU LATHAM 47 BP 1 - CAUDEBEC-EN-CAUX - 76490 RIVES-EN-SEINE Tél. 33 (0) 2 35 56 35 00 - Fax 33 (0) 2 35 56 35 56						5. Work order/Contract/Invoice Bon de commande / Contrat / Facture 07516674 / R413017	
6. Item Item	7. Description Description	8. Part No. N° de pièce	9. Qty Qté	10. Serial No. N° de série	11. Status / Work Etat / Travaux		12. Remarks Remarques		
1	APU APS500G14	4504113B	1	SP-E0314550	REPAIRED				
L1: TSN:14461.4 CSN:18550 TSR:1.2 CSR:4 L2: REPAIRED i.a.w EM 40C14-1 Rev.18 Dated July 01,2022 and L3: CIR 40C14-2 Rev.16 Dated July 01,2022. L4: AD/5B embodied: Refer to REVIMA FOR-R-173 SO: 5464810. L5: Work performed : Refer to Shop Finding Report SO:5464810. L6: DMM reading at departure : 14461.4 hours / 18550 starts. L7: APU preserved for 18 months i.a.w. SPM 490001 Rev5 L8: TASK 49-00-01-620-802 starting on September 29,2023. L9: Log Book delivered with the APU. The work identified in Block 11 and described herein has been accomplished in accordance with 14 CFR part 43 and in respect to that work, the items are approved for return to service under certificate n° CNF5912C. TCCA APPROVAL N°898-35									
13a. Certifies that the items identified above were manufactured in conformity to : Certifie que les éléments identifiés ci-dessus ont été fabriqués conformément aux : ☐ approved design data and are in condition for safe operation données de conception approuvées et sont en état de fonctionner en toute sécurité. ☐ non-approved design data specified in block 12 données de conception non approuvées spécifiées dans la case 12									
13b. Authorised Signature Signature autorisée		13c. Approval/Authorisation Number Numéro d'agrément / d'autorisation		14a. ☒ Part-145.A.50 Release to Service Approbation pour remise en service selon Partie 145.A.50 Certifies that unless otherwise specified in block 12, the work identified in block 11 and described in block 12, was accomplished in accordance with Part-145 and in respect to that work the items are considered ready for release to service. Certifie que, sauf indication contraire spécifiée en case 12, les travaux identifiés en case 11 et décrits en case 12 ont été réalisés conformément à la Partie 145 et qu'au vu de ces travaux, les pièces sont considérées prêtes à la remise en service.		14b. Authorised Signature Signature autorisée		14c. Certificate/ Approval Ref. No. N° du Certificat / Agrément FR.145.0011	
13d. Name Nom		13e. Date (dd/mm/yyyy) Date (jj mmm aaaa)		14d. Name Nom Q.C. APRS # RV0429 Benoit BAL		14e. Date (dd/mm/yyyy) Date (jj mmm aaaa) 29 SEP 2023			
Note : 1) USER/INSTALLER RESPONSABILITIES: this certificate does not automatically constitute authority to install the item(s). Responsabilités de l'utilisateur/l'installateur/ ce document ne constitue pas forcément l'autorisation d'installer l(es) item(s). 2) Where the user / installer performs work in accordance with regulations of an Airworthiness Authority different than the Airworthiness Authority specified in block 1, that his / her Airworthiness Authority accepts items from the Airworthiness Authority specified in block 1. Quand l'utilisateur / installateur travaille selon les réglementations d'une autorité de navigabilité différente de l'autorité de navigabilité mentionnée dans la case 1, que son autorité de navigabilité accepte les items libérés par l'autorité de navigabilité mentionnée dans la case 1. 3) Statements in blocks 13a and 14a do not constitute installation certification. In all cases aircraft maintenance records must contain an installation certification issued in accordance with the national regulation by the user / installer before the aircraft may be flown. Les indications portées en cases 13a et 14a ne constituent pas une certification de montage. Dans tous les cas, le dossier d'entretien de l'aéronef doit contenir une certification d'installation délivrée conformément aux règlements nationaux par l'utilisateur / installateur avant que l'aéronef puisse voler.									

1. Approving Competent Authority/Country CAA/UK		2. AUTHORISED RELEASE CERTIFICATE CAA FORM 1			3. Form Tracking Number 87564 SO: 5464810 / 1.000		
4. Organisation Name and Address: REVIMA 1, AVENUE DU LATHAM 47 BP 1 - CAUDEBEC-EN-CAUX - 76490 RIVES-EN-SEINE Tél. 33 (0) 2 35 56 35 00 - Fax 33 (0) 2 35 56 35 56							
6. Item		7. Description		8. Part No	9. Qty.	10. Serial No	11. Status/Work
1		APU APS500C14		4504113B	1	SP-E0314550	REPAIRED
12. Remarks L1: TSN:14461,4 CSN:18550 TSR:1.2 CSR:4 L2: REPAIRED i.a.w EM 40C14-1 Rev.18 Dated July 01,2022 and L3: CIR 40C14-2 Rev.16 Dated July 01,2022. L4: AD/SB embodied: Refer to REVIMA FOR-R-173 SO: 5464810. L5: Work performed : Refer to Shop Finding Report SO:5464810. L6: DMM reading at departure : 14461,4 hours / 18550 starts. L7: APU preserved for 18 months i.a.w. SPM 490001 Rev5 L8: TASK 49-00-01-620-802 starting on September 29,2023. L9: Log Book delivered with the APU.							
13a. Certifies that the items identified above were manufactured in conformity to: ☐ approved design data and are in a condition for safe operation ☐ non-approved design data specified in block 12				14a. ☒ Part 145.A.50 Release to Service ☐ Other regulation specified in block 12 Certifies that unless otherwise specified in block 12, the work identified in block 11 and described in block 12, was accomplished in accordance with Part 145 and in respect to that work the items are considered ready for release to service.			
13b. Authorised Signature		13c. Approval/Authorisation Number		14b. Authorised Signature 		14c. Certificate/Approval Ref. No UK.145.01553	
13d. Name		13e. Date		14d. Name Q.C. APRS # RV0429 Benoit BAL		14e. Date 29 SEP 2023	
USER/INSTALLER RESPONSIBILITIES This certificate does not automatically constitute authority to install the item(s). Where the user/installer performs work in accordance with regulations of an airworthiness authority different than the airworthiness authority specified in block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts items from the airworthiness authority specified in block 1. Statements in blocks 13a and 14a do not constitute installation certification. In all cases aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.							

STATUS OF LIFE LIMITED PARTS AT DEPARTURE

Customer R/O	R413017
Revima APU S/O	5464810

APU TYPE	APS 500-C14	P/N	S/N	SP-E0314550
TSN	14 461,4	TSO	N/A	TSR
CSN	18 550	CSO	N/A	CSR
				4

DESCRIPTION	P/N	S/N	TSN	CSN	Life limit	Cycles remaining	Origin	Status at departure
TURBINE WHEEL	4504847	HP330	1,2	4	15 000	14 996	NEW	New
COMPRESSOR WHEEL	4504253	SIF8734	1,2	4	50 000	49 996	NEW	New

Completed by: (Stamp & Sign)	B.BAL	29/09/2023
Date:		

1. Approving Civil Aviation Authority/Country: FAA/UNITED STATES	2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG PT10019NE	3. Form Tracking Number: 23552517336
4. Organization Name and Address: PRATT & WHITNEY 3625 ROYAL SOUTH PKWY ATLANTA, GA 30349 US	5. Work Order/Contract/ Invoice Number: 552517336 Page 1 of 2	

6. Item	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status/Work
1	WHEEL, TURBINE	4504847	15	HP290 HP295 HP301 HP304 HP310 HP312 HP314 HP326 HP327 HP330	NEW

12. Remarks: AIRWORTHINESS APPROVAL TSO: C77A ETSO: EASA.IM.210.10017321
This WHEEL is a subcomponent of a TSO authorization. This article meets the special requirements of Canada. Complied with all applicable Airworthiness Directives and Service Bulletins (including Alert Service Bulletins). For a complete list of eligibilities, reference the attached Bilateral Agreement Eligibility Attachment document.

13a. Certifies the items identified above were manufactured in conformity to: ☒ Approved design data and are in a condition for safe operation. ☐ Non-approved design data specified in Block 12.		14a. ☒ 14 CFR 43.9 Return to Service. ☐ Other regulation specified in Block 12. Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.	
13b. Authorized Signature: <i>Shadaisha M. Jones</i>	13c. Approval/Authorization No: ODA800004NE	14b. Authorized Signature:	14c. Approval/Certificate No:
13d. Name(Typed or Printed): SHADAISHA JONES	13e. Date(dd/mm/yyyy): 21/AUG/2023	14d. Name(Typed or Printed):	14e. Date(dd/mm/yyyy):

User/Installer Responsibilities:
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Block 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance record must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

The conditions and tests required for TSO approval of this article are minimum performance standards. It is the responsibility of those installing this article either on or within a specific type or class of aircraft to determine that the aircraft installation conditions are within the TSO standard. TSO articles must be approved for installation per the applicable airworthiness requirements.
FAA Form 8130-3(02-14)

NSN: 0052-00-012-9005

Copie Certifiée - Certified True Copy

Timbre - Stamp: 801 069793/1/2

Nom - Name: KAH BISSOON

Signature: *KAH BISSOON*

Date: 24 AUG 2023

Quantity/Quantity: 15

est expédié avec un certificat de copie certifiée/
is shipped under this certified true copy certificate.

27/9/23

7

7

HP330

Butt

W

RV 160

1. Approving Civil Aviation
Authority/Country:

FAA/UNITED STATES

2.

AUTHORIZED RELEASE CERTIFICATE

FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG

3. Form Tracking
Number:

23552517336

4. Organization Name and Address:

PRATT & WHITNEY

3625 ROYAL SOUTH PKWY

ATLANTA, GA 30349 US

PT10019NE

SOUTHERN LOGISTICS CENTER

5. Work Order/Contract/
Invoice Number:

552517336

Page 2 of 2

6. Item 7. Description:

6. Item	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status/Work
1	WHEEL, TURBINE	4504847	CONTD	HP332 HP342 HP357 HP360 HP363	NEW

12. Remarks (Cont):

13a. Certifies the items identified above were manufactured in conformity to:

- ☒ Approved design data and are in a condition for safe operation.
☐ Non-approved design data specified in Block 12.

13b. Authorized Signature:

13d. Name (Typed or Printed):

SHADAISHA JONES

13c. Approval/Authorization No:

ODA800004NE

13e. Date (dd/mm/yyyy):

21/AUG/2023

14a. ☐ 14 CFR 43.9 Return to Service ☒ Other regulation specified in Block 12
Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and
described in Block 12 was accomplished in accordance with Title 14, Code of Federal
Regulations, part 43 and in respect to that work, the items are approved for return to
service.

14b. Authorized Signature:

14d. Name (Typed or Printed):

14c. Approval/Certificate No:

14e. Date (dd/mm/yyyy):

User/Installer Responsibilities:

It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Block 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance record must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

The conditions and tests required for TSO approval of this article are minimum performance standards. It is the responsibility of those installing this article either on or within a specific type or class of aircraft to determine that the aircraft installation conditions are within the TSO standard. TSO articles must be approved for installation per the applicable airworthiness requirements.

FAA Form 8130-3(02-14)

NSN: 0052-00-012-9005

1. Approving Civil Aviation Authority/Country: FAA/UNITED STATES		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG PT10019NE		3. Form Tracking Number: 22545562626	
4. Organization Name and Address: PRATT & WHITNEY 3825 ROYAL SOUTH PKWY ATLANTA, GA 30349 US		5. Work Order/Contract/ Invoice Number: 545562626		Page 1 of 2	
6. Item Description: 1 IMP ASSY T40C14		8. Part Number: 4504253		9. Quantity: 14	
10. Serial Number: SIF8679 SIF8685 SIF8686 SIF8689 SIF8697 SIF8729 SIF8731 SIF8732 SIF8733 SIF8734		11. Status/Work: NEW			
12. Remarks: AIRWORTHINESS APPROVAL TSO: C77A ETSO: EASA.M.210.10017321 This IMP ASSY T40C14 is a subcomponent of a TSO authorization. This article meets the special requirements of Canada. Complied with all applicable Airworthiness Directives and Service Bulletins (including Alert Service Bulletins). For a complete list of eligibilities, reference the attached Bilateral Agreement Eligibility Attachment document.					
13a. Certifies the items identified above were manufactured in conformity to: ☒ Approved design data and are in a condition for safe operation. ☐ Non-approved design data specified in Block 12.					
13b. Authorized Signature: <i>[Signature]</i>		13c. Approval/Authorization No: ODA800004NE		13d. Approval/Signature No:	
13e. Name/Title or Position: MYRA MITCHELL		13f. Date/Issued/YYMM: 09/AUG/2022		13g. Date/Issued/YYMM:	
It is important to understand that the existence of this document alone does not automatically constitute approval to enter the aircraft engine/propeller/rotor/blade. Where the user/owner performs work in accordance with the national regulations of an airworthiness authority different from the airworthiness authority of the country specified in Block 1, it is essential that the user/owner obtains approval from the airworthiness authority of the country specified in Block 1. Statements of approval from the airworthiness authority of the country specified in Block 1 are required for the user/owner to be able to operate the aircraft engine/propeller/rotor/blade. The conditions and data required for TSO approval of this article are minimum performance standards. It is the responsibility of those installing this article either on or within a specific type or class of aircraft to determine that the article meets the applicable airworthiness requirements.					

NSN: 0052-00-012-9005

Copie Certificate - Certified True Copy

Timbre - Stamp: 200751999

Nom - Name: KAT BISSON

Signature: *[Signature]*

Date: 21/08/2022

Quantity/Quantity: 3

Unit expédition: 3

is shipped under this certified true copy certificate:

Photocopy checked: 27/09/23
Autre d'un document: 1
As a document: 3
Sur une quantité: 3
Par: *[Signature]*
Visa: *[Signature]*
RV 6068

1. Approving Civil Aviation Authority/Country: FAA/UNITED STATES	2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG PT10019NE	3. Form Tracking Number: 22645562626
4. Organization Name and Address: PRATT & WHITNEY 3625 ROYAL SOUTH PKWY ATLANTA, GA 30349 US	SOUTHERN LOGISTICS CENTER	5. Work Order/Contract Invoice Number: 545562626 Page 2 of 2

6. Item	7. Description	8. Part Number	9. Quantity	10. Serial Numbers	11. Status/Work
1	IMP ASSY T40C14	4504253	CONTD	SIF8735 SIF8737 SIF8738 SIF8739	NEW

12. Remarks (Cont):

13a. Certifies the items identified above were manufactured in conformity to:

- ☒ Approved design data and are in a condition for safe operation.
☐ Not approved design data specified in Block 12.

13b. Authorized Signature:

[Signature]

13c. Approval/Authentication No:

QDA800004NE

13d. Name (Typed or Printed):

MYRA MITCHELL

13e. Date (dd/mm/yyyy):

09/AUG/2022

13f. Date (dd/mm/yyyy):

User/Installer Responsibilities:
 It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/airframe where the user/installer performs work in accordance with the national regulations of any airworthiness authority. It is the responsibility of the user/installer to ensure that the airworthiness authority, through the aircraft engine/propeller/airframe, from the airworthiness authority of the country specified in Block 1. Statements in Block 13a and 13b do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification based in accordance with the national regulations by the user/installer before the aircraft may be flown.

The conditions and tests required for TSO approval of this article are minimum performance standards, it is the responsibility of those building this article either on or within a specific type or class of aircraft to determine that the aircraft installation conditions are within the TSO standard. TSO articles must be approved for installation per the applicable airworthiness requirements.

FAA Form 8130-3(02-14)

NSN: 0052-00-012-9005

bc20

Date : 29 sept. 2023

Engine type : APS500 C14
 Serial Number : 0314550
 I.A. Number : 07516674
 Nucleus S/N :
 Gearbox S/N :
 Doc ref : EM 40C14-1 R18 du 01/07/22
 L.H.V. : 18582

AUXILIARY POWER UNIT TEST CELL RECORD SHEET
 =====

	Units	No Load	Bleed air load	Shaft load	Shaft & bleed air load	----- ----- -----
Baro pressure	psi	14.721	14.736	14.721	14.736	0.000
Inlet Temperature T1	oF	66.5	68.0	66.7	68.5	0.0
Bleed air flow	lb/min	-----	86.82	-0.08	69.60	0.00
Cor. bleed air flow	lb/min	-----	85.66	-0.08	23.06	0.00
Compressor pressure	psi	-----	43.23	55.88	45.92	0.00
Load Pressure	psi	-----	53.461	14.953	58.436	0.000
Load Pressure cor	psi	-----	58.19	-----	60.58	0.00
Bleed air temp.	oF	-----	440.4	146.8	440.4	0.0
Cor. bleed air temp.	oF	-----	440.4	-----	440.6	0.0
Engine EGT 1	oF	591.6	1258.1	637.3	1098.1	0.0
Engine EGT 2	oF	628.3	1245.3	666.3	1083.5	0.0
Tail pipe EGT	oF	591.4	1274.9	623.6	1095.0	0.0
Cor. tail pipe EGT	oF	-----	1253.6	0.0	1106.4	0.0
Turbine vibration	G	3.30	5.07	5.21	4.14	0.00
Accessory vibration	G	0.59	1.06	1.54	1.26	0.00
Tail pipe pressure	psi	14.34	14.53	14.35	14.51	0.00
Oil pressure	psi	67.64	67.79	67.93	67.71	0.00
Oil temperature	oF	185.5	215.9	203.5	214.8	0.0
Gearbox air pres.	psi	-0.07	-0.05	-0.07	-0.05	0.00
Fuel temperature	oF	65.8	65.3	65.4	65.4	0.0
fuel pressure	psi	23.84	18.59	23.59	17.10	0.00
Corrected shaft load	kW	0.0	0.0	18.5	10.2	0.0
Fuel flow	lb/h	100.7	190.2	110.6	173.4	0.0
Cor. fuel flow	lb/h	100.5	189.7	110.4	174.3	0.0
N1 speed	rpm	63438	63438	63477	63477	0
Hourmeter	hours	14460.2	-----	-----	-----	14461.4
Event counter	cycles	18546	-----	-----	-----	18550
Start time	sec.	23.0	-----	-----	-----	-----
Cool down time	sec.	-----	-----	-----	-----	22.0

Operating limits

Min. air flow	lb/min	-----	81.1	-----	-----	-----
Max. E.G.T corrected	oF	-----	-----	-----	1199.0	-----
Min. Press load cor	psi	-----	51.27	-----	56.28	-----
Max cor. load temp.	oF	-----	473.3	-----	473.7	-----
Max cor. fuel flow	lb/h	-----	-----	-----	183.6	-----

Technician

RV 6238

Control

RV 352A

MODIFICATION STATUS AT DEPARTURE

Customer AJ WALTER USD (V) A/C type Regionnal A/C A/C # N704PG Customer PO R413017
 APU type APU APS500C14 APU P/N 4504113A APU S/N SP-E0314550 Revima SO 5464810
 TSN 14461,4 TSO UNK TSR 1,2 Revima WO 7516674
 CSN 18550 CSO UNK CSR 4

NCW: Not Complied With PCW: Previously Complied With CW: Complied With NR: Not Removed
 NA3: Not Applicable due to PN NA4: Not Applicable due to SN NA5: Part missing CL: SB CANCELLED

DATE AND VISA

29 SEP 2023

RV 0429

Type	Rev	SB nb	Cat	Summary	Status
AD	R00	AD2004-24-03	A	Introduction of Bracket to Prevent Fuel Filter Button Protusion	NA3
ASB	R01	4504112-49-26	A	Inspection of Duplex Fuel Nozzle Assembly Part Number 4951830	PCW
ASB	R03	4504112-49-22	A	Introduction of Bracket to Prevent Fuel Filter Button Protusion	NA3
RSIL	R00	APS500R-001	B	Replacement of tube P/N 4504727	CW
SB	R01	T-62T-49-58	B	Magnetic Chip Detector Inspection Procedure	NA3
SB	R02	RSI-091	D	Modification of air inlet housing to apply abradable coating more resistant to	NCW
SB	R01	4504616-49-2	B	Modification of Fuel Control Assy with improved crack resistance	CW
SB	R00	4504112-49-8	B	Introduction of Quill Shaft to Prevent Interference with Turbin Bearin	NA4
SB	R02	4504112-49-6	D	Intruduction of Fuel Control Assembly, Part Number 4504616A	PCW
SB	R00	4504112-49-5	B	Introduction of Engine Control Harness Assembly Part Number 4504230A	PCW
SB	R00	4504112-49-45	B	Application of the run-on or prevailing torque of the V-band clamps	CW
SB	R00	4504112-49-44	D	Introduction of new bleed valve assembly	NCW
SB	R00	4504112-49-43	A	Removing reference to the life management plan	CW
SB	R00	4504112-49-42	D	Replacement of Turbine Retainer Assy by P/N 4550159	NCW
SB	R00	4504112-49-40	D	Data Memory Module - Part Number update	NA3
SB	R00	4504112-49-39	F	Introduction of GEARBOX HOUSING ASSEMBLY	NCW
SB	R02	4504112-49-38	B	Introduction of turbine wheel retainer	CW
SB	R00	4504112-49-36	B	Introduction of oil pump seal shield	CW
SB	R00	4504112-49-35	D	Inspection of Air Inlet Duct Assembly Shims	NCW
SB	R00	4504112-49-34	D	Introduction of Servo-Controlled Bleed Air Valve, Part Number 4952829	PCW
SB	R01	4504112-49-32	B	Introduction of Speed Sensor to Reduce APU Auto Shutdown and Speed-Rel	PCW
SB	R00	4504112-49-30	C	Introduction of Air Inlet Housing Assembly with Improved Moisture and	PCW
SB	R00	4504112-49-29	B	Introduction of Combustor Housing Assembly with Improved Baffle	PCW
SB	R00	4504112-49-27	D	Cold Start Bypass Valve Modif. to Prevent Low Oil Pressure	PCW
SB	R00	4504112-49-24	B	Incorporation of Modif. Plate with Additionnal Identifier	NA4
SB	R02	4504112-49-23	B	Introduction of Modified Duplex Fuel Nozzle Assemblies	PCW
SB	R00	4504112-49-21	C	Introduction of Air Inlet Housing Assembly to Reduce Stress-Induced Cr	PCW
SB	R03	4504112-49-20	D	Introduction of High Oil Temperature Switch with Improved Over-Tempera	PCW
SB	R01	4504112-49-19	D	SB Superseded by SB 4504112-49-34	PCW
SB	R02	4504112-49-17	D	Introduction of Remotely-Mounted Low Oil Pressure Switch	PCW
SB	R02	4504112-49-15	D	Introduction of Reinforced Drain Line To Prevent Cracking	NA5
SB	R01	4504112-49-14	D	Introduction of Flexible Bleed Duct Assembly Part Number 4952281	NA4
SB	R01	4504112-49-13	D	De-Icing Fluid Resistant Fuel Solenoid Valve	PCW
SB	R01	4504112-49-11	D	APU Modif. to Part Number M4504113A	NA4
SB	R02	4504112-49-10	D	Introduction d'une bleed valve P/N 4952304	PCW
SB	R02	4504112-49-1	D	Replacement of Part Number 4951627 Fuel Filter Assembly with Part Numb	PCW

End of document

LRU Inventory at departure

APU P/N 4504113A Customer AJ WALTER TUA (WCS)
APU S/N SP-E0314550 Customer PO R413017
APU Type APU APS500C14 Revima SO 5464810

APU TSN: 14460,2
APU CSN: 18546

At receipt			At departure								
Part Number	Serial Number	Description	Part number	Serial number	TSN	TSO	TSI	CSN	CSO	CSI	Workscope
4952316-2	2727	VALVE, SOLENOID-FUEL	4952316-2	2727	UNK	UNK	UNK	UNK	UNK	UNK	Continuous time
4951486	0704	FLOW DIVIDER	4951486	0704	UNK	UNK	UNK	UNK	UNK	UNK	Continuous time
4950745	03052895	EXCITER, IGNITION	4950745	J17440513	UNK	UNK	0	UNK	UNK	0	Serviceability review
4951491	03148387	CABLE, IGNITION	4951491	03148387	UNK	UNK	UNK	UNK	UNK	UNK	Continuous time
4950899	2004	VALVE, ANTI-SURGE	4950899	2121	UNK	UNK	0	UNK	UNK	0	Repaired
4952829	3480	VALVE, BLEED AIR-SERVO CONTROL	4952829	3480	UNK	UNK	UNK	UNK	UNK	UNK	Continuous time
4504196A	V04-559020	AMBIENT PRESSURE(UNAMPLIFIED) & DATA MEMORY MODULE ASSEMBLY	4504196A	V04-559020	14460,2	UNK	UNK	18546	UNK	UNK	Continuous time
4505576	0649	SPEED SENSOR ASSY, DUAL HALL E	4505576	0649	UNK	UNK	UNK	UNK	UNK	UNK	Continuous time
4504185C	087637	SWITCH, HIGH OIL TEMPERATURE	4504185C	087637	UNK	UNK	UNK	UNK	UNK	UNK	Continuous time
4952397-1	0302	INDICATOR PRESSURE DIFFERENTIAL	4952397-1	0302	UNK	UNK	UNK	UNK	UNK	UNK	Continuous time
179349-1	2519	SWITCH, LOW OIL PRESSURE	179349-1	2519	UNK	UNK	UNK	UNK	UNK	UNK	Continuous time
4952366	495	FUEL CONTROL ASSEMBLY	4952366	495	UNK	UNK	0	UNK	UNK	0	Repaired
4504616A	REV500C14-002	NOZZLE ASSY, DUPLEX FUEL	4504616B	REV500C14-002	UNK	UNK	0	0	0	0	Supplied part
M4951830	1NT07789	NOZZLE ASSY, DUPLEX FUEL	4952404	W6572	0	0	0	0	0	0	Supplied part
M4951830	1TT07816	NOZZLE ASSY, DUPLEX FUEL	4952404	W6638	0	0	0	0	0	0	Supplied part
M4951830	1X508758	NOZZLE ASSY, DUPLEX FUEL	4952404	W6194	0	0	0	0	0	0	Supplied part
M4951830	1TT04338	NOZZLE ASSY, DUPLEX FUEL	4952404	W6688	0	0	0	0	0	0	Supplied part
M4951830	1OT13342	NOZZLE ASSY, DUPLEX FUEL	4952404	W6635	0	0	0	0	0	0	Supplied part
M4951830	1NV02322	NOZZLE ASSY, DUPLEX FUEL	4952404	W6571	0	0	0	0	0	0	Supplied part
4951609	M87709-01	MANIFOLD ASSEMBLY-PRIMARY FUEL	4951609	M87709-01	14460,2	UNK	0	18546	UNK	0	Serviceability review
4951610	M82225-01	MANIFOLD ASSEMBLY-SECONDARY FU	4951610	M82225-01	14460,2	UNK	0	18546	UNK	0	Repaired
4952039	C0066	FILTER ASSEMBLY, FUEL	4952505	C2050	0	0	0	0	0	0	Supplied part
160544-1	N/A	IGNITOR	160544-1	N/A	0	0	0	0	0	0	Supplied part
160544-1	N/A	IGNITOR	160544-1	N/A	0	0	0	0	0	0	Supplied part
4950929	737430-08	COOLER AIR/OIL	4950929	737430-08	UNK	UNK	UNK	UNK	UNK	UNK	Continuous time
4504230B	V28-5829	HARNESS ASSY, ENGINE CONTROL	4504230B	V28-S829	UNK	UNK	0	UNK	UNK	0	Repaired
4950581-1	N/A	CABLE IGNITION	4950581-1	N/A	0	0	0	0	0	0	Supplied part

The blue bag had to be split due to the certification and teardown report being sealed inside this. Have done a visual inspection of the unit as well as possible due to the container.

I DID CLARIFY WITH A 145 INSPECTOR BEFORE I SPLIT THE BAG.



M

DANIEL SCHLOTTER.
