1. Approving Civil Aviation Authority/Country:  FAA/United States 4. Organization Name and Address:		THINESS APPROVAL TAG	3. Form Tracking Number: W61449222511  W616449222511  5. Work Order, Contract or Invoice Number:
075	10990 NW 92nd Terrace MIAMI, FL 33178 Ph: 305-482-9903, Fax: 305-482-9905 sales@generalmroaerospace.com		5. Work Order, Contract or Invoice Number:
	FAA Approval Holder: 19GR994Y		46766514
o. Item: /. Description:	8. Part Number:	9. Quantify:	10 Serial Number:
ACTUATOR	251A4510-10	or mountility.	KE01249
I C DEMARNO.			
Full details describing the actual work perform. This unit was Overhauled in accordance with: MFG: BOEING ATA Manual ID: 27-45-12, Rev	Full details describing the actual work performed are contained and held in W/O # W61449. This unit was Overhauled in accordance with: MFG: BOEING ATA Manual ID: 27-45-12, Revision #: 57, Revision Date: 7/1/2023	61449.	
Complied with AD 2011-27-03 27-45-12, Rev. 57, DTD July 0	Complied with AD 2011-27-03 by accomplishing SB 737-27A1277, Rev. 2, DTD January 08, 2010. Unit has been Modified from P/N: 251A4510-9 to P/N 251A4510-10 IAW Boeing CMM ATA 27-45-12, Rev. 57, DTD July 01, 2023.	ary 08, 2010. Unit has been Modified from P/N: 25:	\4510-9 to P/N 251A4510-10
Certifies that the work specified under EASA Part-145 Approval	Certifies that the work specified in block 11/12 was carried out in accordance with EASA Part-145 and in respect to that work the component is considered ready for release to service under EASA Part-145 Approval Number: "EASA.145.6095".	SA Part-145 and in respect to that work the compor	nt is considered ready for rel
Approved design data and are	Approved design data and are in a condition for safe operation.  Non-approved design data specified in Block 12	14a. X 14 CFR 43.9 Return to Service X Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service	X Other regulation specified in Block 12 Block 12, the work identified in Block 11 and accordance with Title 14, Code of Federal work, the items are approved for return to service.
405			
on a superior of the superior	13c. App. wal Authorization No:	14b. Authorized Signature:	14c. Approval/Certificate No. 19GR994Y
(Types of Fillian).	13e. Date (dd/mmm/yyyy):	14d. Name (Typed or Printed):	14e. Date (dd/mmm/yyyy):
		Adam Supersor	31/Oct/2023
t is important to understand that the o	User/Insta	User/Installer Responsibilities	
Where the user/installer work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country user/installer ensures that his/her airworthiness authority accepts aircraft engine/shortest by the country user/installer ensures that his/her airworthiness authority accepts aircraft engine/shortest by the country of the country user/installer ensures that his/her airworthiness authority accepts aircraft engine/shortest by the country of the country	xistence of this document alone does not automatically		
	lance with the national regulations of an airworthiness or thiness authority accepts aircraft engine(s)/propeller(	<ul> <li>constitute authority to install the aircraft engine/pr authority different than the airworthiness authority (s)/article(s) from the airworthiness authority of the</li> </ul>	peller/article.
Statement in Blocks 13a and 14a do no egulations by the user/installer before	Where the user/installer work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1 it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.  Statement in Blocks 13a and 14a do not constitute installation certification. In all cases aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.	constitute authority to install the aircraft engine/prauthority different than the airworthiness authority (s)/article(s) from the airworthiness authority of the maintenance records must contain an installation	peller/article.  f the country specified in Blo ountry specified in Block 1. certification issued in accord



General MRO Aerospace 10990 NW 92nd Terrace MIAMI, FL 33178

Ph: 305-482-9903, Fax: 305-482-9905 sales@generalmroaerospace.com

<u>Teardown</u>

Work Order #: W61449

Date Printed: 10/31/2023

Time: 9:33:24 AM

Page: 1

To: BOEING THE BOEING COMPANY PO BOX 3707 SEATTLE WA 98124-2207 USA.

Ship To: BOEING DALLAS SERVICE CENTER 2750 REGENT BOULEVARD SHIPPING RAMP S2 DFW AIRPORT, TX 75261 USA

Ref #: 46766514

Phone #: 206-662-6650

Fax #:

Email: matthew.m.vaji@boeing.com

Code: B1

Site: 1EL

PN: 251A4510-10

Descr: ACTUATOR

Qty: 1

phase Schiller

Make:

Mfg: BOEING

Serial Number: KE01249

Entry Date: 4/24/2023

**Need Date:** 

Compl. Date: 10/30/2023 4:

Warranty: No

Container:

ATA: 27-45-12 Loc: 2728 Revision: 57 Rev Date: 7/1/2023

REMARKS

TSI: UNK TSN: 48342.48 CSI: UNK CSN: 27248

SUBCOMPONENT PN AND SN

Ball nut and Screw Assy P/N: 07322P000-05 S/N: 1275

REASON FOR REMOVAL

Scheduled.

## PRELIMINARY INSPECTION

Verified P/N and S/N from RO and unit. Unit arrived dirty, external contamination and corrosion, paint and cadmium damaged, leaking oil from the secondary brake housing, corrosion on the upper gimbal, missing bolts and washers on guard, corroded bearing on the support assy, deep wear marks and pitting on screw shaft, broken lock-wire hole on the cover of upper housing, missing hardware, missing pin, and hardware.

#### **PRE TEST**

Unit was not tested due to incoming condition. Oil leakage upon arrival.

### HIDDEN DAMAGE

No hidden damage reported.

# **TEARDOWN FINDINGS**

During inspection the following discrepancies were found:

Unit was cleaned and inspected. Unit was not tested due to incoming condition. Unit was fully disassembled for further inspection. Upon inspection found excessive internal contamination, compressed packings, rings and seals, missing lower pins and hardware, missing nuts from cable guard, worn screw shaft, corroded and worn bearings, over heating bearings on primary brake, worn bushings, damaged paint and cadmium plate, and broken wire hole on the cover housing. Unit requires all mentioned parts, 100% kit, deep cleaning, NDT, CAD plating, surface treatment, paint, and reconditioning for Overhaul.

Discrepancies Noted:

1. Worn screw shaft. Requires replacement.

Maintenance Release Statement (FAR Pars 121, 125 and 135)

The instruments and or Components above was Serviced and Inspected with Pride and in accordance with current F.A.A. Regulations and is | X| Approved | | Not approved to Return to Service. Pertinent details of the Services are on files at this Repair Station.

Name:

Authorized Signature:

Date:

10/31/2023

Adam Superson

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GMA Form 340



General MRO Aerospace 10990 NW 92nd Terrace MIAMI, FL 33178

Ph: 305-482-9903, Fax: 305-482-9905 sales@generalmroaerospace.com

Teardown

Work Order #: W61449

Date Printed: 10/31/2023

Time: 9:33:25 AM

Page: 2

To: BOEING THE BOEING COMPANY PO BOX 3707 SEATTLE WA 98124-2207 USA,

Ship To: BOEING DALLAS SERVICE CENTER 2750 REGENT BOULEVARD SHIPPING RAMP S2 DFW AIRPORT, TX 75261 USA

Ref #: 46766514

Phone #: 206-662-6650

Fax #:

Email: matthew.m.vaji@boeing.com

Code: B1

Site: 1EL

PN: 251A4510-10

Descr: ACTUATOR

Qty: 1

Make:

Mfg: BOEING

Serial Number: KE01249

Entry Date: 4/24/2023

**Need Date:** 

Compl. Date: 10/30/2023 4:1

Warranty: No

Container:

ATA: 27-45-12 Loc: 2728 Revision: 57 Rev Date: 7/1/2023

- 2. Missing lower pin and hardware. Requires replacement.
- 3. Worn and corroded bearings. Requires replacement.
- Overheating bearings. Requires replacement.
- Broken wire hole on the cover housing. Requires I/H rework.
- 6. Cadmium plating faded. Requires CAD plating.
- 7. Internal contamination. Requires deep cleaning.
- 8. Compressed packings and seals. Requires replacement.
- 9. Damaged paint. Requires paint.

#### Rework Required:

- 1. Deep cleaning.
- Surface treatment.
- 3. NDT.
- 4. I/H rework.
- 5. CAD plating.
- 6. Paint.
- 7. Reconditioning

# AD/SERVICE BULLETINS/MODIFICATIONS

Complied with AD 2011-27-03 by accomplishing SB 737-27A1277, Rev. 2, DTD January 08, 2010. Unit has been Modified from P/N: 251A4510-9 to P/N 251A4510-10 IAW Boeing CMM ATA 27-45-12, Rev. 57, DTD July 01, 2023.

## WORK PERFORMED

Performed replacement of all mentioned parts, deep cleaning, surface treatment, NDT, I/H rework, CAD plating, assembly, and testing. Unit passed all functional tests. Performed paint and final reconditioning. Unit has been Overhauled IAW Boeing CMM ATA 27-45-12, Rev. 57, DTD July 01, 2023.

Maintenance Release Statement (FAR Pars 121, 125 and 135)

The instruments and or Components above was Serviced and Inspected with Pride and in accordance with current F.A.A. Regulations and is | X| Approved | | Not approved to Return to Service. Pertinent details of the Services are on files at this Repair Station.

Name:

Authorized Signature:

Date:

10/31/2023

Adam Superson

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