USER/INSTALLER RESPONSIBILITIES		13a. Certifies that the items identified above were manufactured in conformity to paper approved design data and are in a condition for safe operation non-approved design data specified in Block 12	12. REMARKS This document has been issued according to an approved computer generated signature procedure of this document has been issued according to an approved computer generated signature procedure OEM Part Number: S360N021-114 The work identified in Block 11 and described herein has been accomplished in accordance with 14 are approved for release or return to service under certificate no. VM8Y694J. Applicable Technical Docs: CMM 77-30-10 REV.2 DATED 03/02/2017 For work carried out, refer to Service Breakdown Report for Service Order 45432554	1 Engine Vibration Monitor Unit (EVMU)	Item Description	dress	SWITZERLAND 2. SWITZERLAND
JEROME METTRAUX	140	14	Instance. In Instance In Insta	32-110	Part No.	Meggitt SA Route de Moncor 4, CH-1752 Villars-sur-Glâne, Switzerland Phone: +41 (0)26 407 11 11 Fax: +41 (0)26 407 18 60	AUTHORISED RELEASE CERTIFICATE EASA FORM 1
	Electronic Signature on File	to Service X C scified in Block 12, the was cordance with Part -145 a srvice	he items	0923	Serial No.	5. Work Order/Contract/Invoice Notification: 350534704 PO: R407662 Item:	
14e. Date (dd mmm yyyy) 20APR2023	14c. Certificate/Approval Ref No. CH.145.0251	a. X Part-145.A.50 Release to Service X Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identifed in Block 11 and described in Block 12 was accomplished in accordance with Part -145 and in respect to that work the items are considered ready for release to service		 INSPECTED	11. Status / Work	ice 14704 362	3. Form Tracking Number 86818513-10300
		id in		0/	Vork		

This certificate does not automatically constitute authority to install the item(s).

Where the user/installer performs work in accordance with regulations of an airworthiness authority different than the airworthiness authority specified in block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts items from the

SER/INSTALLER RESPONSIBILITIES		approved design data and are in a condition for safe operation non-approved design data specified in Block 12 a. Authorised Signature	OEM Part Number: S360N021-114 Applicable Technical Docs: CMM 77-30-10 REV.2 DATED 03/02/2017 For work carried out, refer to Service Breakdown Report for Service Order 45432554	This document has been issued popular.		- 1	tem Description		4. Organisation Name and Address	1. Approving Competent Authority/Country CAA/UK 2.
JEROME METTRAUX	VM OS COA DY	14	computer generated signature procedure. 03/02/2017 rr Service Order 45432554			Part No. Qty	0	Meggitt SA Route de Moncor 4, CH-1752 Villars-sur-Glâne, Switzerland Phone: +41 (0)26 407 11 11 Fax: +41 (0)26 407 18 60		AUTHORISED RELEASE CERTIFICATE CAA FORM 1
14e. Date (dd mmm yyyy) 20APR2023	Electronic Signature on File UK.145.01715	a. X Part-145.A.50 Release to Service Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identifed in Block 11 and described in Block 12 was accomplished in accordance with Part -145 and in respect to that work the items are considered ready for release to service			0923	Serial No.		5. Work Order/Contract/Invoice Notification: 350534704 PO: R407662 Item:		3. Form Tracking Number 86818513-10300
	Ref No.	in Block 12 and described in k the items are			INSPECTED / TESTED	Status / Work				

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MEGGÍTT

Service Breakdown Report

Sold to
A.J.WALTER AVIATION
The Headquarters
Maydwell Avenue, Slinfold
WEST SUSSEX
RH13 0AS
UNITED KINGDOM

 Page
 1 of 1

 Date
 20.03.2023

 Service notification
 350534704

Meggitt quality notification

 Service order/Report No
 45432554

 Sales order
 9953463223

 Customer PO#
 R407662

Customer quality ref Customer Part#

Equipment serviced here

Item Part number/ Serial number Mod Status/Revision
Description

1 241-258-032-110 0923

EVM 258 - Engine Vibration Monitor Unit Applicable Technical Docs:

CMM 77-30-10 REV.2 DATED 03/02/2017

Failure / complaint (as reported by customer)

for test: under t&m - reason for removal: no cert A/C Reg: @@9Y-ANU@@ TSN: 57023 CSN: 23307

for stock

EASA FAA CAA

Findings

Service Rep EVM-GENERIC

Self test pass.

No fault in memory.

Unit works within specification.

No fault found.

Materials

Item Qty UOM Part Number Batch Number Description Cond

Work performed

Incoming Inspection (Admin)

Preliminary hidden damage inspection

SB/SIL/AD status check performed

Visual Inspection

Check connectors condition

Read & Analyze Internal Memory

Incoming ATP Technician

Incoming ATP Test System

Tested & Inspected IAW CMM listed above

Final ATP Technician

Final ATP Test System

Final FOD check

Administration

BH (VX)-P)

TEST REPORT

Subject:

ACCEPTANCE TEST REPORT

Item:

Engine Vibration Monitoring Unit

Type:

EVM-258

MFR:

S3960

PNR:

241-258-032-110

Cust. PNR:

S360N021-114

Mod. Record:

.

SER:

0923

DMF:

072000

Applicable Doc.:

ATP-1141-10 rev-02

Software PNR:

299-500-000-015

Database PNR:

299-527-500-102

AU SER:

AC85211

Operator:

B.Hildebrand

Test Result:

PASS

Started:

09 Mar 2023 10:24:47

Finished:

09 Mar 2023 10:52:53

Report generated:

09 Mar 2023 10:52:53

Report Location:

\\ngts\atr\atp1141_10r02_evm258/P-0923-20230309-105253.html

GTS Hostname:

gtsxp0011

Comment:

Signature:

By (VM-P)

VM-QE

Department Approval:

0 9 MARS 2023

Date of Approval: